

Aprox. constants for exam purposes: $M_{\oplus} = 6 \cdot 10^{24} \text{ kg}$, $R_{\oplus} = 6 \cdot 10^6 \text{ m}$, $G = \frac{2}{3} \cdot 10^{-10} \text{ Nm}^2 \text{ kg}^{-2}$, $c = 3 \cdot 10^8 \text{ ms}^{-1}$

Plotting graph copies provided. Use multiple copies if necessary. Most problems are solved more easily by geometry.

EXAM SHEET 1: First do this exam with closed book in 2 hours. EXAM SHEET 2: Re-do this exam open source no time limit.

Both parts of this Assignment 15 due by Thursday Dec. 14

Io returns (With a vengeance!)

1. Suppose *Io* volcano **L** ejects lava at such a speed $v_{IN45^\circ}^{Hit90^\circ}$ that ejecta with initial angle $\alpha=45^\circ$ have range $\rho=90^\circ$ to hit equatorial target T_{90° . (On graph of *Io* assume mass $M_{\oplus}=6 \cdot 10^{24} \text{ kg}$ and radius $R_{\oplus}=6 \cdot 10^6 \text{ m}$ near to that of Earth. Use R_{\oplus} as a geometric unit. This *Io* has an R_{\oplus} -circular-orbit of period $t_{\oplus} = \underline{\hspace{2cm}}$ min.)

a. First find and label “focus-locus” line for fixed $\alpha=45^\circ$ and varying v_{IN} and locate 2nd focal point (or points) corresponding to pole-to-equator flight.

b. Construct this T_{90° orbit while indicating major & minor axes, foci, and enough points to show its shape. Plot KE/PE ratio R -scale and show eccentricity vector ϵ and give its magnitude $|\epsilon| = \epsilon = \underline{\hspace{2cm}}$.

c. From this estimate its initial KE/PE ratio $R = \underline{\hspace{2cm}}$ and initial $v_{IN45^\circ}^{Hit90^\circ} = \underline{\hspace{2cm}} \text{ km/s}$.

d. Estimate the time to impact $t_{L\text{-to-}T_{90^\circ}} (\alpha=45^\circ) = \underline{\hspace{2cm}}$ minutes. (Derive and discuss geometrically using Kepler laws.)

e. Construct and label the “focus-locus” circle for fixed $v_0 = v_{IN45^\circ}^{Hit90^\circ}$ and varying α . Does another initial path with the same $v_0 = v_{IN?}^{Hit90^\circ}$ but different angle $\alpha = \underline{\hspace{2cm}}?$ also hit T_{90° ? If so find or construct its orbital geometry (major & minor axes, foci, and ϵ), value of KE/PE ratio $R = \underline{\hspace{2cm}}$, and time to impact $t_{L\text{-to-}T_{90^\circ}} = \underline{\hspace{2cm}}$.

f. What v_0 -trajectory has minimum time to impact $t_{MIN-L\text{-to-}T_{90^\circ}}$ (Give $\alpha_{MIN} = \underline{\hspace{2cm}}^\circ$ and $t_{MIN-L\text{-to-}T_{90^\circ}} = \underline{\hspace{2cm}}$ min.)

More difficult problems:

g. Construct an envelope contacting all trajectories with the same initial speed v_0 but various launch angles α .

Indicate the contact points of the envelope with ellipse(s) drawn so far.

h. Tell if targets T_{30° __?, T_{60° __?, T_{120° __?, T_{150° __?, T_{180° __?, are also within range. (Yes, No, or maybe)

i. Can missiles launched at $\alpha=45^\circ$ with higher $v_{IN} > v_0$ reach beyond T_{90° to T_{120° __?, to T_{180° __?, beyond T_{180° __?

j. Discuss what ellipse, parabola, or hyperbola limits $\alpha=45^\circ$ -path(s) to a maximum range $(\rho_{max})^\circ = \underline{\hspace{2cm}}$ (what?)

k. Construct points of that maximal trajectory visible on the graph (include path below $R_{\oplus} = 6 \cdot 10^6 \text{ m}$) while we now imagine actual mass of *Io* has been compressed to a Schwarzschild black hole of radius $r_{\oplus S}$ at C.

(Xtra cred: If instead it is a neutron starlet plowing thru *Io*, would it deviate above or below the compressed *Io*-path? Discuss)

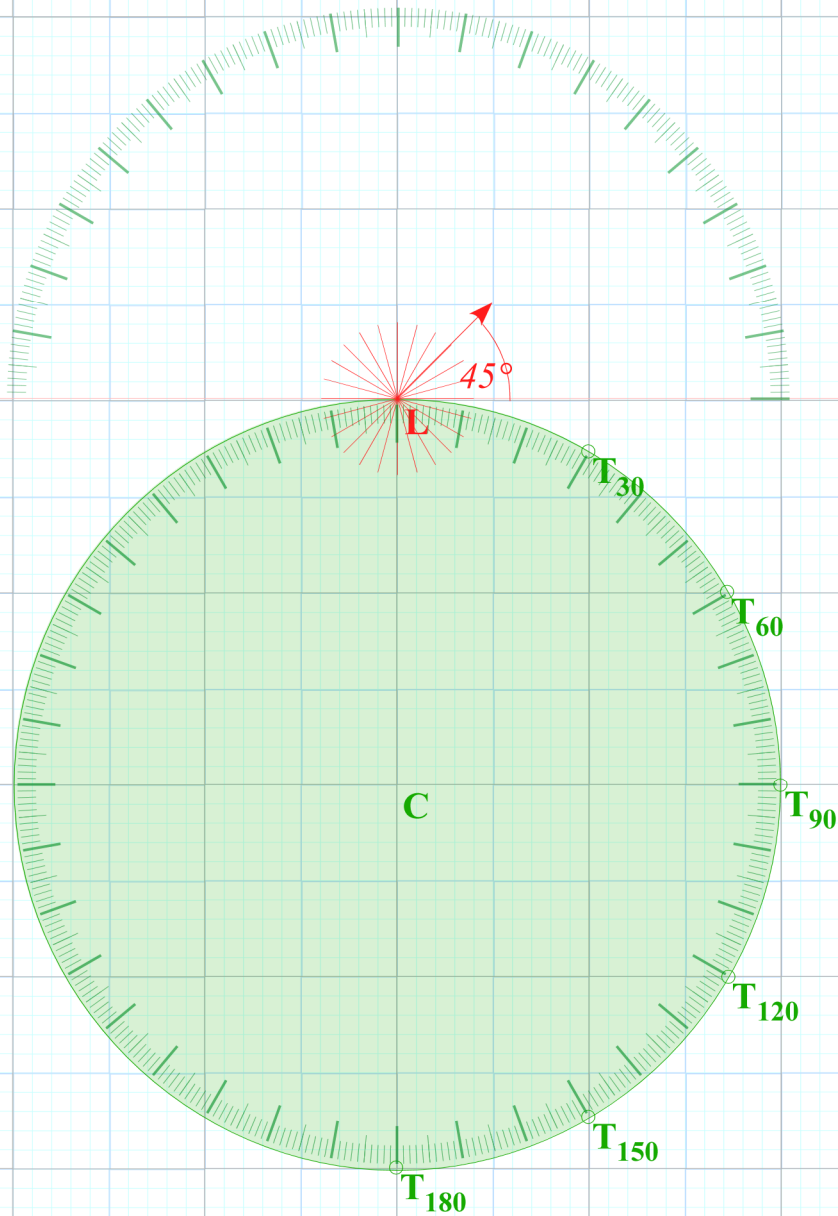
l. About how big is a Schwarzschild-limited Earth radius? $r_{\oplus S} = \underline{\hspace{2cm}}$ cm. Plot $r_{\oplus S}$ if possible.

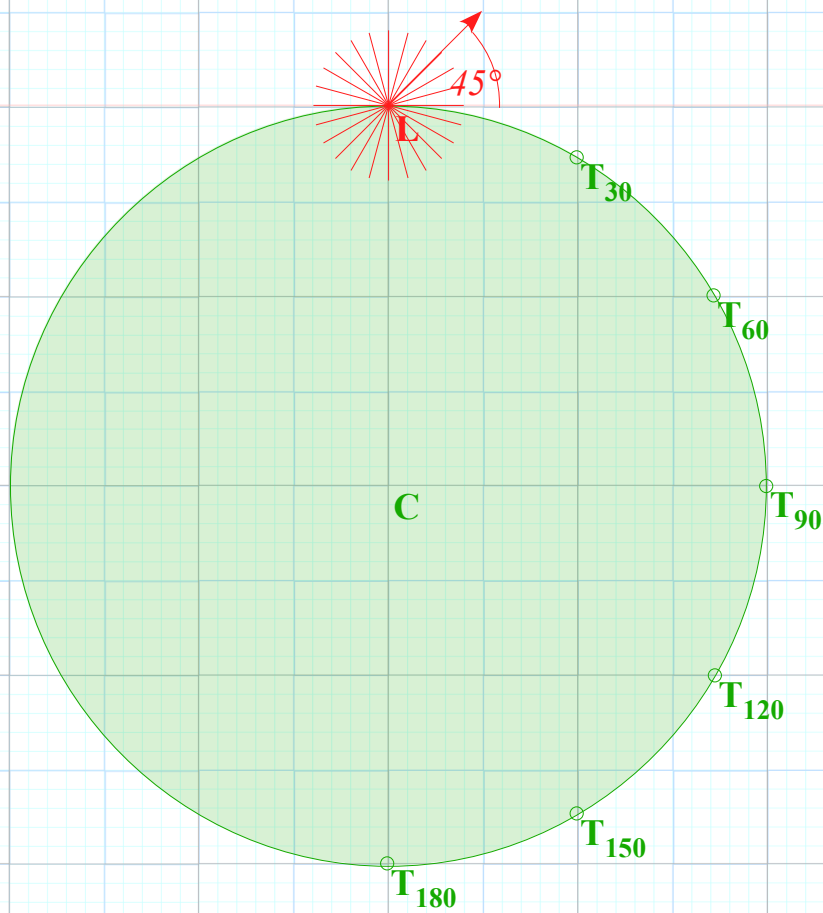
Still more difficult problems:

m. Find $v_{IN} < v_0$ -trajectory of minimum KE/PE ratio R_{MIN} to hit T_{90° (Give $\alpha_{MINR} = \underline{\hspace{2cm}}^\circ$, $\epsilon = \underline{\hspace{2cm}}$, and $R_{MIN} = \underline{\hspace{2cm}}$)

Sketch or construct orbit. Derive algebraic formulae if possible.

n. Give radius r_R of circular fixed- R “focus-locus” in terms of initial radius r_o and ratio R . $r_R = \underline{\hspace{2cm}}$





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a. First find and label “focus-locus” line for fixed $\alpha=45^\circ$ and varying v_{IN} and locate 2nd focal point (or points) corresponding to pole-to-equator flight.

b. Construct this **T** $_{90^\circ}$ orbit while indicating major & minor axes, foci, and enough points to show its shape. Plot *KE/PE* ratio *R*-scale and show eccentricity vector ϵ and give its magnitude $|\epsilon| = \epsilon = \underline{\hspace{2cm}}$.

c. From this estimate its initial *KE/PE* ratio $R = \underline{\hspace{2cm}}$ and initial $v_{IN45^\circ}^{Hit90^\circ} = \underline{\hspace{2cm}} \text{ km/s}$.

d. Estimate the time to impact $t_{L\text{-to-}T_{90^\circ}} (\alpha=45^\circ) = \underline{\hspace{2cm}}$ minutes. (Derive and discuss geometrically using Kepler laws.)

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f. What v_0 -trajectory has minimum time to impact $t_{MIN\text{-L-to-}T_{90^\circ}}$ (Give $\alpha_{MIN} = \underline{\hspace{2cm}}^\circ$ and $t_{MIN\text{-L-to-}T_{90^\circ}} = \underline{\hspace{2cm}}$ min.)

More difficult problems:

g. Construct an envelope contacting all trajectories with the same initial speed v_0 but various launch angles α . Indicate the contact points of the envelope with ellipse(s) drawn so far.

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i. Can missiles launched at $\alpha=45^\circ$ with higher $v_{IN} > v_0$ reach beyond **T** $_{90^\circ}$ to **T** $_{120^\circ}$ __?, to **T** $_{180^\circ}$ __?, beyond **T** $_{180^\circ}$ __?

j. Discuss what ellipse, parabola, or hyperbola limits $\alpha=45^\circ$ -path(s) to a maximum range (ρ_{max})^o = _____ (what?)

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